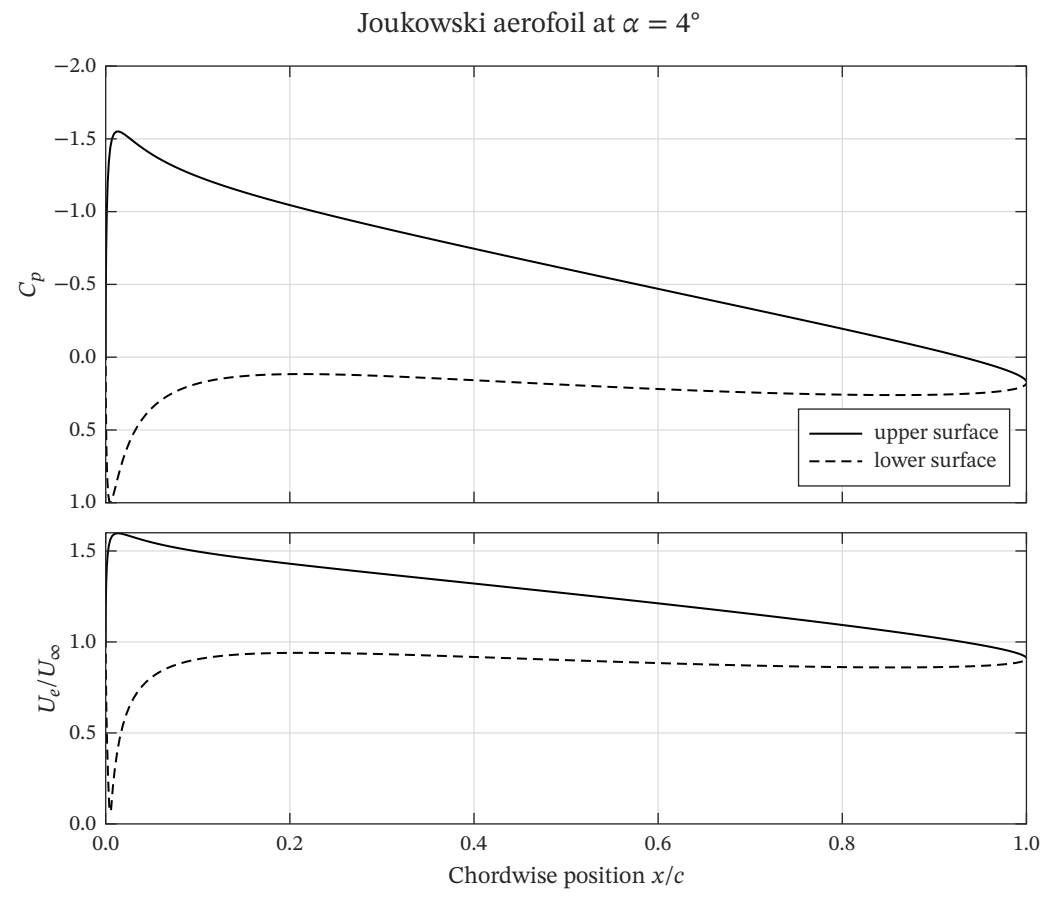

Joukowski aerofoil at $\alpha = 4°$
C_p
−2.0
−1.5
−1.0
−0.5
0.0
0.5
1.0
upper surface
lower surface
U_e/U_∞
0.0
0.5
1.0
1.5
0.0
0.2
0.4
0.6
0.8
1.0
Chordwise position x/c